

## Dissipation-induced instabilities and symmetry

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**Abstract** The paradox of destabilization of a conservative or non-conservative system by small dissipation, or Ziegler's paradox (1952), has stimulated a growing interest in the sensitivity of reversible and Hamiltonian systems with respect to dissipative perturbations. Since the last decade it has been widely accepted that dissipation-induced instabilities are closely related to singularities arising on the stability boundary, associated with Whitney's umbrella. The first explanation of Ziegler's paradox was given (much earlier) by Oene Bottema in 1956. The aspects of the mechanics and geometry of dissipation-induced instabilities with an application to rotor dynamics are discussed.

**Keywords** Dissipation-induced instabilities · Destabilization paradox · Ziegler pendulum · Whitney umbrella

### 1 Introduction

There is a fascinating category of mechanical and physical

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systems which exhibit the following paradoxical behavior: when modeled as systems without damping they possess stable equilibria or stable steady motions, but when small damping is introduced, some of these equilibria or steady motions become unstable [1–5]. A systematic survey of the literature was given by Kirillov and Verhulst [6].

The paradoxical effect of damping on dynamic instability was noticed first for rotor systems which have stable steady motions for a certain range of speed, but which become unstable when the speed is changed to a value outside the range. In 1924, Kimball studied the destabilization of a flexible rotor in stable rotation at a speed above the critical speed for resonance [7]. In fact, in 1879 Thomson and Tait [8] showed already that a statically unstable conservative system which has been stabilized by gyroscopic forces could be destabilized again by the introduction of small damping forces. However, the destabilization by damping, using Routh's theorems, is implicit in their calculations, it is not formulated as a paradox.

### 2 Ziegler's paradox

In 1952, Hans Ziegler of ETH Zurich published a paper [9] that became classical and widely known in the community of mechanical engineers; it also attracted the attention of mathematicians. Ziegler was interested in flutter problems in aerodynamics and considered a double pendulum, fixed at one end and compressed by a tangential end load. He unexpectedly encountered a phenomenon with a paradoxical character: the domain of stability of the Ziegler's pendulum changes in a discontinuous way when one passes from the case in which the damping is very small to that where it has vanished [9,10].

In the conclusion to his classical book, Bolotin emphasized that the discrepancy between the stability domains of undamped non-conservative systems and that of systems with infinitesimally small dissipation is a topic of the greatest theoretical interest in stability theory which is shown in Ref. [11]. Encouraging further research of the destabilization paradox, Bolotin was not aware that the crucial ideas for its explanation were formulated by Bottema as early as 1956 [12]. Surprisingly, this paper surpassed the attention of most scientists during five decades.

### 3 Bottema opened Whitney’s umbrella

In a remarkable paper of 1943 [13], Hassler Whitney described singularities of maps from  $\mathbb{R}^n$  into  $\mathbb{R}^m$  with  $m = 2n - 1$ . It turns out that in this case a special kind of singularity plays a prominent role. Later, the local geometric structure of the manifold near the singularity has been aptly called Whitney’s umbrella. In Fig. 1, the original sketch of the singular surface from the companion article [14] is reproduced. The basic idea is this. Consider a  $n$ -dimensional manifold with a singularity at the origin. The manifold is mapped into  $m$ -space with  $m = 2n - 1$ . To be concrete, assume  $n = 2, m = 3$ , the simplest interesting case. In a neighborhood of the origin it is possible to find coordinates such that we have exactly

$$y_1 = x_1^2, \quad y_2 = x_2, \quad y_3 = x_1x_2, \tag{1}$$

so that  $y_1 \geq 0$  and on eliminating  $x_1$  and  $x_2$

$$y_1y_2^2 - y_3^2 = 0. \tag{2}$$

Starting on the  $y_2$  axis for  $y_1 = y_3 = 0$ , the surface widens up for increasing values of  $y_1$ . For each  $y_2$ , the cross-section is a parabola; as  $y_2$  passes through 0, the parabola degenerates to a half-ray, and opens out again (with sense reversed) in Fig. 1.

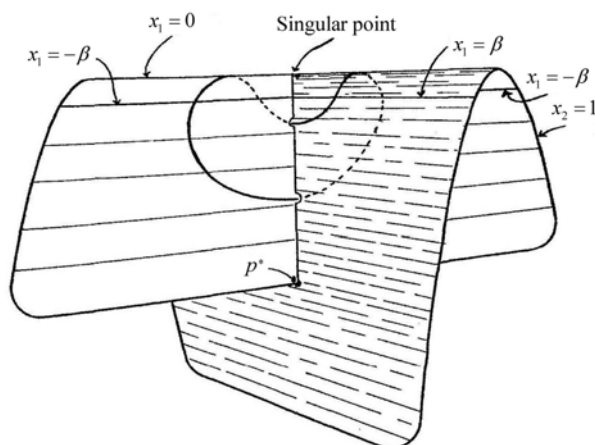


Fig. 1 Whitney’s original 1944 sketch of the umbrella

The analysis on singularities of functions and maps has become a fundamental ingredient for bifurcation studies of differential equations. After the pioneering work of Hassler Whitney and Marston Morse, it has become a huge research field, both in theoretical investigations and in applications. In 1943, it was hard to imagine that this study of global analysis, a pure mathematical abstraction, would find already an industrial application in the next decade.

### 3.1 Bottema’s solution

In 1956, there appeared an article by Oene Bottema (1901-1992), that outstripped later findings for decades [12]. Bottema’s work in 1955 can be seen as an introduction, it was directly motivated by Ziegler’s paradox [15]. In 1956, he considered a much more general class of small oscillations of non-conservative systems near the equilibrium configuration  $x = y = 0$

$$\begin{aligned} \ddot{x} + a_{11}x + a_{12}y + b_{11}\dot{x} + b_{12}\dot{y} &= 0, \\ \ddot{y} + a_{21}x + a_{22}y + b_{21}\dot{x} + b_{22}\dot{y} &= 0, \end{aligned} \tag{3}$$

where  $a_{ij}$  and  $b_{ij}$  are constants.

The matrices  $\mathbf{A} = (a_{ij})$  and  $\mathbf{B} = (b_{ij})$  can both be written uniquely as the sum of symmetrical and antisymmetrical parts:  $\mathbf{A} = \mathbf{K} + \mathbf{N}$  and  $\mathbf{B} = \mathbf{D} + \mathbf{G}$ , where

$$\begin{aligned} \mathbf{K} &= \begin{pmatrix} k_{11} & k_{12} \\ k_{21} & k_{22} \end{pmatrix}, & \mathbf{N} &= \begin{pmatrix} 0 & \nu \\ -\nu & 0 \end{pmatrix}, \\ \mathbf{D} &= \begin{pmatrix} d_{11} & d_{12} \\ d_{21} & d_{22} \end{pmatrix}, & \mathbf{G} &= \begin{pmatrix} 0 & \Omega \\ -\Omega & 0 \end{pmatrix}, \end{aligned} \tag{4}$$

with  $k_{11} = a_{11}, k_{22} = a_{22}, k_{12} = k_{21} = (a_{12} + a_{21})/2, \nu = (a_{12} - a_{21})/2$  and  $d_{11} = b_{11}, d_{22} = b_{22}, d_{12} = d_{21} = (b_{12} + b_{21})/2, \Omega = (b_{12} - b_{21})/2$ .

The characteristic equation for the frequencies of the small oscillations around equilibrium is

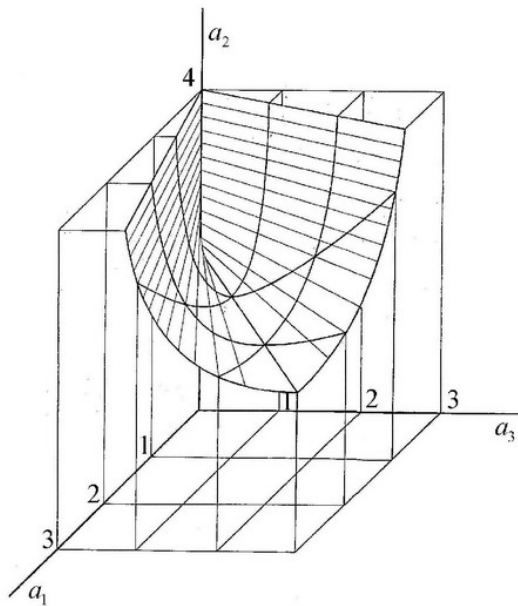
$$Q := \lambda^4 + a_1\lambda^3 + a_2\lambda^2 + a_3\lambda + a_4 = 0, \tag{5}$$

where the parameters  $a_1, a_2, \dots, a_4$  depend on the 8 parameters of the system. For the equilibrium to be stable all roots of the characteristic equation need to have real parts which are non-positive, pure imaginary multiple eigenvalues have to be semi-simple.

After a long linear algebra analysis, it was found that the condition for the surface  $V$  that separates stable and unstable motion is

$$a_1a_2a_3 = a_1^2a_4 + a_3^2. \tag{6}$$

When  $a_4 = 1$ , which can be assumed without loss of generality, this is the equation of a ruled surface  $V$  of the third degree, which the condition for  $a_1 \geq 0, a_3 \geq 0$  should be considered, see Fig. 2. It is equivalent to Whitney’s umbrella in the case  $n = 2, m = 3$ . Note that when starting off with 8 parameters in the system, but that the surface  $V$  bounding the stability domain is described by 3 parameters.



**Fig. 2** Original drawing from the 1956 work [12] of Oene Bottema, showing the domain of the asymptotic stability of the two-dimensional non-conservative system with Whitney's umbrella singularity. The ruled surface (called  $V$  in the text) is given by Eq. (6)

### 3.2 The role of symmetry

Interestingly, Bottema's analysis in 1956 also shows that the ratio of the damping coefficients is important. Explicitly, in the case of symmetric damping,  $b_{11} = b_{22}$ , the destabilization is not very effective. It is shown by a simple example. Consider the system

$$\begin{aligned} \ddot{x} + x + y + \kappa_1 \dot{x} &= 0, \\ \ddot{y} - x + \omega^2 y + \kappa_2 \dot{y} &= 0, \end{aligned} \quad (7)$$

with damping coefficients ( $\kappa_1, \kappa_2 \geq 0$ ). The characteristic equation for the eigenvalues becomes

$$(\lambda^2 + \kappa_1 \lambda + 1)(\lambda^2 + \kappa_2 \lambda + \omega^2) + 1 = 0. \quad (8)$$

Without damping,  $\kappa_1 = \kappa_2 = 0$ , the trivial solution is unstable if  $0 < \omega^2 < 3$  and stable if  $\omega^2 > 3$ . In the case of stability, the eigenvalues are purely imaginary. If  $\omega^2 = 3$ , it is so-called Krein-collision.

The eigenvalues without and with damping for  $\omega^2 = 4$  using MATLAB are presented. Without damping, it is

$$\omega^2 = 4, \quad \kappa_1 = \kappa_2 = 0,$$

eigenvalues:  $1.9021i, 1.1756i$ .

A type of asymmetric damping:  $\kappa_1 > 0, \kappa_2 = 0$ .

$$\omega^2 = 4, \quad \kappa_1 = 0.1,$$

eigenvalues:  $-0.0585 \pm 1.1736i, +0.0085 \pm 1.9029i$ ;

$$\omega^2 = 4, \quad \kappa_1 = 0.2,$$

eigenvalues:  $-0.1164 \pm 1.1678i, +1.0164 \pm 1.9053i$ .

Damping in the first degree of freedom ( $x$ ) destabilizes. Now symmetric damping

$$\omega^2 = 4, \quad \kappa_1 = \kappa_2 = 0.1,$$

eigenvalues:  $-0.05 \pm 1.9015i, -0.05 \pm 1.1745i$ ;

$$\omega^2 = 4, \quad \kappa_1 = \kappa_2 = 0.2,$$

eigenvalues:  $-0.10 \pm 1.8995i, -0.10 \pm 1.1713i$ .

In the case of symmetric damping, it is not necessary to be destabilization.

For a general system (3) without gyroscopic forces ( $\Omega = 0$ ) with the matrices (4) stability boundary equation (6) reads  $\nu = \nu_{cr}$ , where for  $\left| \frac{2\text{tr}KD - \text{tr}K\text{tr}D}{2\text{tr}D} \right| \ll \nu_0$  we can approximately write [7]

$$\nu_{cr} \approx \nu_0 - \frac{1}{2\nu_0} \left[ \frac{2\text{tr}KD - \text{tr}K\text{tr}D}{2\text{tr}D} \right]^2, \quad (9)$$

with

$$\nu_0^2 := \left( \frac{\text{tr}K}{2} \right)^2 - \det K. \quad (10)$$

In case of no damping, the non-conservative system (3) without gyroscopic forces is stable for  $\nu^2 < \nu_0^2$ . In the presence of dissipation for its asymptotical stability, it is necessary that  $\nu^2 < \nu_{cr}^2$ . Obviously, when the damping matrix is proportional to a unit matrix  $I$  so that  $D = \kappa I$ , it follows from Eq. (9) that  $\nu_{cr}^2 \approx \nu_0^2$ . Therefore, in the limit of vanishing damping the system with symmetrical damping is stable at the same range of variation of the parameter  $\nu$  as that without damping. The symmetrical damping stabilizes the system.

In general, non-symmetrical damping (with arbitrary matrix  $D = D^T$ ) destabilizes the system because according to Eq. (9), there is  $\nu_{cr}^2 \leq \nu_0^2$ . However, a class of non-symmetric damping matrices  $D$  such that [16]

$$2\text{tr}KD - \text{tr}K\text{tr}D = 0, \quad (11)$$

is stabilizing again.

If  $D$  depends on two parameters, say  $\delta_1$  and  $\delta_2$ , then Eq. (9) has a canonical form equation (2) for Whitney's umbrella in the  $(\delta_1, \delta_2, \nu)$ -space. Due to discontinuity existing for  $2\text{tr}KD - \text{tr}K\text{tr}D \neq 0$ , the equilibrium may be stable if there is no damping, but unstable if there is damping, however small it may be. Also the critical non-conservative parameter,  $\nu_{cr}$ , depends on the ratio of the damping coefficients is observed and thus it is strongly sensitive to the distribution of damping.

## 4 Parametric resonance in systems with dissipation

Parametric resonance arises usually in applications if there is an independent (periodic) source of energy. The classi-

cal example is the mathematical pendulum with oscillating support and a typical equation studied in this context is the Mathieu equation, see Fig. 3a for this classical case. In applications with parametric excitation where usually more degrees of freedom play a part, many combination resonances are possible. In what follows, the so-called sum resonance will be important.

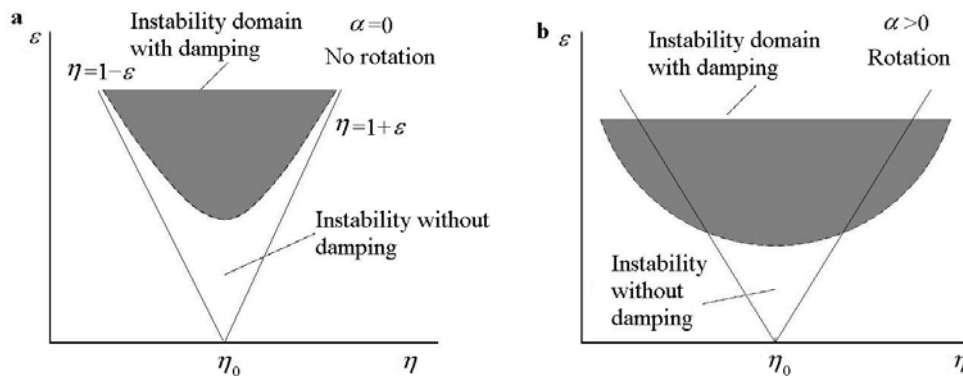
### 4.1 Rotor dynamics without damping

The effects of adding linear damping to a parametrically excited system have already been observed and described in for instance Ref. [11]. The following example is based on Ref. [17], see also Ref. [18]. Consider a rigid rotor consisting of a heavy disk of mass  $M$  which is rotating with constant rota-

tion speed around an axis. The axis of rotation is elastically mounted on a foundation; the connections which are holding the rotor in an upright position are also elastic. Assuming small oscillations in the upright position, frequency  $2\eta$ , the equations of motion without damping become after rescaling

$$\begin{aligned} \ddot{x} + 2\alpha\dot{y} + (1 + 4\epsilon\eta^2 \cos 2\eta t)x &= 0, \\ \ddot{y} - 2\alpha\dot{x} + (1 + 4\epsilon\eta^2 \cos 2\eta t)y &= 0. \end{aligned} \tag{12}$$

System (12) constitutes a conservative system of coupled Mathieu-like equations. The parameter  $\alpha$  is proportional to the rotation speed  $\Omega$ . The natural frequencies of the unperturbed system (12),  $\epsilon = 0$ , are  $\omega_1 = \sqrt{\alpha^2 + 1} + \alpha$  and  $\omega_2 = \sqrt{\alpha^2 + 1} - \alpha$ . It is shown in Ref. [17], using complex variables, that we can transform this system to two identical Mathieu equations.



**Fig. 3** **a** The classical case as we find for instance for the Mathieu equation with and without damping; in the case of damping the instability tongue is lifted off from the  $\eta$ -axis and the instability domain is reduced. **b** The instability tongues for the rotor system. Again, because of damping, the instability tongue is lifted off from the  $\eta$  axis, but the tongue broadens. The boundaries of the V-shaped tongue without damping are to first approximation described by the expression  $\eta = \sqrt{1 + \alpha^2}(1 \pm \epsilon)$ ,  $\eta_0 = \sqrt{1 + \alpha^2}$

Using the classical and well-known results for the Mathieu equation, it is concluded that the trivial solution is stable for  $\epsilon$  small enough, provided that  $\sqrt{\alpha^2 + 1}$  is not close to  $n\eta$ , for  $n = 1, 2, \dots$ . The first-order and most prominent interval of instability,  $n = 1$ , arises if  $\sqrt{\alpha^2 + 1} \approx \eta$ . Note that this instability arises when

$$\omega_1 + \omega_2 = 2\eta,$$

i.e. when the sum of the eigenfrequencies of the unperturbed system equals the excitation frequency  $2\eta$  which is the sum resonance of first order. The domain of instability is bounded by

$$\eta_b = \sqrt{1 + \alpha^2} (1 \pm \epsilon) + O(\epsilon^2). \tag{13}$$

See Fig. 3b where the V-shaped instability domain is presented in the case of rotor rotation ( $\alpha \neq 0$ ) without damping. Higher order combination resonances can be studied in the same way; the domains of instability in parameter space con-

tinue to narrow as  $n$  increases.

### 4.2 Rotor dynamics with damping

Small linear damping to system (12) is added, with positive damping parameter  $\mu = 2\epsilon\kappa$ . This leads to the equations

$$\begin{aligned} \ddot{x} + 2\alpha\dot{y} + (1 + 4\epsilon\eta^2 \cos 2\eta t)x + 2\epsilon\kappa\dot{x} &= 0, \\ \ddot{y} - 2\alpha\dot{x} + (1 + 4\epsilon\eta^2 \cos 2\eta t)y + 2\epsilon\kappa\dot{y} &= 0. \end{aligned} \tag{14}$$

Because of the damping term, reduce the system to two identical second order real equations are no longer reduced, as they are done previously. To calculate the instability interval around the value  $\eta_0 = \frac{1}{2}(\omega_1 + \omega_2) = \sqrt{\alpha^2 + 1}$ , the normal form or (periodic solution) perturbation theory is applied, see Ref. [17] for details, to find for the stability boundary

$$\eta_b = \sqrt{1 + \alpha^2} \left( 1 \pm \epsilon \sqrt{1 + \alpha^2 - \frac{\kappa^2}{\eta_0^2}} + \dots \right)$$

$$= \sqrt{1 + \alpha^2} \left( 1 \pm \sqrt{(1 + \alpha^2)\varepsilon^2 - \left(\frac{\mu}{2\eta_0}\right)^2} + \dots \right). \quad (15)$$

It follows that, as in other examples, the domain of instability actually becomes larger when damping is introduced, see Fig. 3b. The instability interval shows a discontinuity at  $\kappa = 0$ . If  $\mu$  or  $\kappa \rightarrow 0$ , then the boundaries of the instability domain tend to the limits  $\eta_b \rightarrow \sqrt{1 + \alpha^2}(1 \pm \varepsilon \sqrt{1 + \alpha^2})$ , which differs from the result it found when  $\kappa = 0$ :  $\eta_b = \sqrt{1 + \alpha^2}(1 \pm \varepsilon)$ . For reasons of comparison, the instability tongues are displayed in Fig. 3 in the four cases with and without rotation, with and without damping.

Mathematically, the bifurcational behavior is again described by Whitney's umbrella as indicated before. In mechanical terms, the broadening of the instability-domain is caused by the coupling between the two degrees of freedom of the rotor in lateral directions which arises in the presence of damping.

## 5 Conclusions

It is remarkable that Bottema's solution in 1956 was ignored for such a long time. For instance Google Scholar gives no citations of the papers in the period of 1956–2008.

The generality of the results described in Sect. 2, enables us to discuss the part played by symmetric damping. One should consult the original 1956 paper to observe the behavior of the eigenvalues with regards to the damping coefficients.

In the context of dissipation-induced instability, the influence of asymmetric and symmetric damping was studied extensively in Refs. [1–5,16]. In these papers Bottema's results were also generalized to higher (more than 4) dimensions. For the rotor system of Sect. 3, the analysis regarding asymmetric damping was carried out in Ref. [5].

Note that the phenomena described here are basically linear and in this sense the dynamics is dominated by the linear terms. Further away from equilibrium and in some critical cases, Krein-collision or small real parts near the umbrella surface, nonlinear terms may come into play [19].

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